

# Horizon 2020-MSCA-RISE

# FATIGUE RESPONSE OF AS-BUILT DMLS MARAGING STEEL AND EFFECTS OF AGING, MACHINING, AND PEENING TREATMENTS

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Consortium

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**UniBo – Experimental mechanics** 



## FATIGUE RESPONSE OF AS-BUILT DMLS MARAGING STEEL AND EFFECTS OF AGING, MACHINING, AND PEENING TREATMENTS



EOSINT M 280



# Introduction

Maraging steel MS1 as a promising materials for AM.High density, hardness up to 50 HRC after the aging heat treatment, comparable to that of wrought material.

High **corrosion resistance** and very high **static strength** (UTS > 2,000 MPa after the aging treat).

**Few studies** on high cycle **fatigue** properties; some mainly dealing with low cycle fatigue or metallurgic matters.

Studies are **missing** regarding Maraging parts in the **"as built"** state. The only available scientific **results** deal with fatigued **Ti-6Al-4V** in the **as built condition**.



# **Motivation and subject**

#### Motivations

- Serious drawbacks may arise from missing post-processing.
- Poor surface finishing → high roughness acting as crack trigger
- High tensile residual stresses arising from the stacking process, which cannot be relaxed without aging.
- No extensive studies in the literature clarify if post-processing treatments, can be skipped and which is the best compromise.

#### **Subject**

Investigating the **joint effects** (and interaction) of **heat treatment**, **machining**, and s**hot-peening** on the fatigue response of DMLS-built Maraging Steel vs. the <u>as-</u><u>fabricated state (as built)</u>.





- Maraging steel MS1 (18% Ni Maraging 300 or AISI 18Ni300). .
- 10-15 samples per combination. •
- Fatigue tests aimed at the determination of the S-N curves in the finite life ۰ domain and of fatigue limits.
- Data have been processed by ISO 12107 (S-N curves) and by the Dixon method for short staircase sequences.
- Statistical analysis. ٠

Set #6

Set #5

Set #4

Set #9

**Results** 1000 Waximum bending stress [MPa] Exp. Data Set #9 Exp. Data Set #8 Exp. Data Set #7 10%, 50%, 90% failure Exp. Data Set #6 probability curves, 90% Exp. Data Set #5 Exp. Data Set #4 confidence level Exp. Data Set #3 ▲ Exp. Data Set #2 Plain material • Exp. Data Set #1 Tests under rotating bending O Run outs R = -1 —S-N curve f = 60Hz - - Lower limit - - Upper limit 100 1.00E+05 1.00F+06 1.00E+07 Life cycles

All aligned results  $\rightarrow$  Global S-N curve interpolated through 56 data.

One the **most complete, general and reliable curves** in the current literature on Maraging Steel.



# **Materials and method**

- Maraging steel MS1 (18% Ni Maraging 300 or AISI 18Ni300).
- Tests under rotating bending (ISO 1143).



#### **Experiment**:

- Heat treatment (ON-OFF)
- Machining: no, yes, subsequent shot-peening

Base plate



# **Materials and method**



- 15 samples per combination.
- Determination of the S-N curves and of fatigue limits.
- Data processed by ISO 12107 and by the Dixon method.
- Statistical analysis.



# **Materials and method**

#### Production

Samples produced by EOSINT M280 system, layer thickness: 40  $\mu m.$ 

#### **Post-manufacture treatments**

- All the samples: micro-shot-peening (stainless steel spherical shots, 400µm diameter, 5 bar flow pressure).
- Sets H and HM: Age-hardening at 490°C for 6 hours.
- Sets M and HM: machining with 0.5 mm allowance.
- Sets MP: machining with 0.5 mm allowance, then shot peening











• Added a fifth treatment: MP

1-factor design

- 15 samples.
- Machining (0.5 mm allowance) and shot-peening (steel shots with 0.7 mm diameter, 5 bar) after machining.
- To take advantage of both **surface finishing** and peening induced **compressive residual stresses**.



# **Experimental procedure**

 All the samples have been checked for dimensions and roughness.

> "As fabricated":  $R_a$ =4.1µm, machined:  $R_a$ =0.5µm, after shot-peening:  $R_a$ =1µm (impact dimples).

- Fatigue tests under four-point rotating bending. Machine by Italsigma, Forlì, Italy.
- Run-out: 10<sup>7</sup> cycles.
- Fractography and micrography by stereoscopic and optical microscopes.









2-by-2 design: effects and interaction of heat treat. and machining.

Fatigue strength remarkably enhanced for Set HM





# Discussion

Heat treatment (alone) is not significant.

Machining (alone) is not significant.

**But** Heat treatment & machining (together) are highly significant  $\rightarrow$  huge interaction.

#### **Possible reasons**

- Heat treatment makes Maraging steel more notch-sensitive.
- Not refined surface → a lot of surface asperities (notches) → even (slightly) detrimental effect.
- Machined ground surface → no asperities → highly effective → great interaction with synergic fatigue behavior enhancement.



#### 1000 **Results** 900 800 700 **One-factor table:** Maximum bending stress [MPa] machining and shot- $\triangle$ peening. Shot-peening after machining seems to have a beneficial impact $10^{4}$ $10^{5}$ $10^{6}$ $10^{7}$ on finite life fatigue. Life Cycles ▲ Exp. Res. Set N -S-N Set #N ▲ Exp. Res. Set M S-N Set M ▲ Exp. Res. Set MP -S-N Set MP

#### Machiningers with subseq. machining

No	Yes	
Set N	Set M	Set MP

Not aged



# Discussion

1-factor ANOVA-based method: differences between the fatigue strengths vs. data scattering (uncertainty, sum of squares of the residuals)

 $SSBC = \left[ \left( \overline{S_N} - \overline{S_{..}} \right)^2 + \left( \overline{S_M} - \overline{S_{..}} \right)^2 + \left( \overline{S_{MP}} - \overline{S_{..}} \right)^2 \right]$ 

Effect of machining

Integral means over the considered life span ( $10^5$  to  $10^7$ )  $\rightarrow$  scalars to be processed in a conventional ANOVA after being scaled by their d.o.f.

**Shot-peening after machining** significantly **improves** the fatigue strength (finite life domain) as an effect of additional **compressive residual stress**.

Shot-peening is conversely **usually recommended before machining** by EOS → original contribution to post-processing optimization.





UTS=2050 MPa, after aging; **but** UTS' is just 1100 MPa, after building (no aging).

Set N: fatigue limit = 426 MPa

21% UTS (aged material)

38% UTS' (not aged material)



# Fractography



Failures started from the surface or, more often, from subsurface porosities of voids.

Lack of penetration, unmolten particles, spots of oxides can be regarded as the **primary failure mechanisms**.



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# **Micrography**



#### Without aging



With aging

Laser scans on the build plane (sample cross section).

Reciprocally angled by 67°C on adjacent layers.

Samples without heat treatment: well visible scanning pattern.

Treated samples: more uniform structure, but scanning traces still visible.



# Conclusion

- Heat treatment and machining, taken alone, do not have a significant effect.
- If applied together, they have a beneficial synergic effect (interaction). Aging treatment recommended on refined surfaces.
- Shot peening after machining has a beneficial effect even without heat treatment. → Recommended protocol for post-processing.
- Samples in the "as built conditions" have a fatigue limit corresponding to 426 MPa (38% UTS' for not aged material).



# **Publications (Journals only)**

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# HORIZON 2020

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Thank you!



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